

5(2): Orbital Precession as a Thomas Precession

The Thomas precession for a rotation of 2π is:

$$\Delta\phi = 2\pi \left(\left(1 - \frac{v^2}{c^2}\right)^{-1/2} - 1 \right) \quad (1)$$

where v is the orbital velocity of the rotation of the metric.

The experimentally observed change is:

$$\Delta\phi = 2\pi \left(\frac{3MG}{ac^2(1-e)} - 1 \right) \quad (2)$$

where M is the mass of the attracting body, G is the Newton constant, a is the semi-major axis of the orbit and e is its eccentricity. The result (2) is obtained from the general method of UFT 403. Before the orbital velocity of the ECE2 metric is given by:

$$\left(1 - \frac{v^2}{c^2}\right)^{-1/2} = \frac{3MG}{ac^2(1-e)} \quad (3)$$

If

$$v \ll c \quad (4)$$

$$\frac{1}{2} \frac{v^2}{c^2} \sim \frac{3MG}{ac^2(1-e)} \quad (5)$$

and

$$\boxed{v^2 \sim \frac{6MG}{a(1-e)}} \quad (6)$$

The ECE2 covariant metric is defined by the infinitesimal line element:

$$ds^2 = c^2 d\tau^2 = c^2 dt^2 - dr^2 - r^2 d\phi^2 \quad (7)$$

in plane polar coordinates. Thus precession is the frame rotation:

$$\phi' = \phi + \omega t \quad (8)$$

1. $v \ll c$
 This frame rotation results in the relativistic angular velocity:

$$v = \omega r \rightarrow (9)$$

$$\Omega = \omega \left(1 - \frac{v^2}{c^2}\right)^{-1/2} \quad (10)$$

and the time dilation:

$$dt' = \left(1 - \frac{v^2}{c^2}\right)^{1/2} dt \quad (11)$$

so

$$\Delta\phi = \Omega dt' - \omega dt$$

$$= 2\pi \left(\left(1 - \frac{v^2}{c^2}\right)^{-1/2} - 1 \right) \quad (12)$$

is one orbit of 2π radians, Q.E.D.

This is any planar orbit of a mass m around a mass M , and any observable precession can be thought of as an ECE2 covariant Thomas precession. Each observable precession has its own v .

The Thomas precession is attributed to the ECE2 variant force law:

$$F = -\nabla\phi + \underline{\omega} \times \underline{p}_0 \quad (13)$$

a self consistent way, because the force law (13) and the line element (7) are both ECE2 covariant. They defined a space of finite torsion and curvature.

Thomas precession is due to the vacuum force:

$$\underline{F}(\text{vac}) = \underline{\omega} \times \underline{p}_0 \quad (14)$$

where $\underline{\omega}$ is the spin angular velocity vector and where ϕ is gravitational potential.

$$\phi_0 = -\frac{MG}{r} \quad (15)$$

As slow in UFT 403 & special relativity produce:

$$\Delta\phi = \frac{r^2}{2} \left(\frac{\omega}{r} - \frac{d\omega}{dr} \right) \quad (16)$$

This is the rotation defined by:

$$\psi \sim \pi \left(1 + \frac{6MG}{c^2 a(1-e)} \right) \quad (17)$$

If there were no precession the angle ψ between the two sides of an ellipse is π . From eq. (17):

$$\Delta\phi := \Delta\phi = \frac{6MG\pi}{c^2 a(1-e)} = 2\pi \left(\frac{3MG}{ac^2(1-e)} \right) \quad (18)$$

as in eq. (2), Q.E.D.

From eqs. (16) and (18):

$$\begin{aligned} \Delta\phi &= \frac{r^2}{2} \left(\frac{\omega}{r} - \frac{d\omega}{dr} \right) = \frac{6\pi MG}{c^2 a(1-e)} \\ &= 2\pi \left(1 - \frac{v^2}{c^2} \right)^{-1/2} \end{aligned} \quad (19)$$

In general:

$$\boxed{\frac{r^2}{2} \left(\frac{\omega}{r} - \frac{d\omega}{dr} \right) = 2\pi \left(1 - \frac{v^2}{c^2} \right)^{-1/2}} \quad (20)$$

So the magnitude of the spin vector is defined by the velocity v of the rotation of the ECE2 covariant metric.

Finally from UFT403:

$$\Delta\psi = \frac{4}{3} \frac{\langle \underline{S}_r \cdot \underline{S}_r \rangle}{r} - \frac{1}{3r} \frac{d}{dr} \langle \underline{S}_r \cdot \underline{S}_r \rangle \quad (21)$$

where: $\frac{d}{dr} \langle \underline{S}_r \cdot \underline{S}_r \rangle \approx \frac{4}{a} \langle \underline{S}_r \cdot \underline{S}_r \rangle \quad (22)$

so

$$\Delta\psi = \frac{4}{3} \frac{\langle \underline{S}_r \cdot \underline{S}_r \rangle}{r} \left(\frac{1}{r} - \frac{1}{a} \right) \quad (23)$$

$$= 2\pi \left(1 - \frac{v^2}{c^2} \right)^{-1/2}$$

This is true in general for any precession.

For gravitational precession:

$$\Delta\psi = \frac{4}{3} \frac{\langle \underline{S}_r \cdot \underline{S}_r \rangle}{r} \left(\frac{1}{r} - \frac{1}{a} \right) = \frac{6\pi MG}{ac^2(1-e)} \quad (24)$$

At the perihelia:

$$r = \frac{d}{1+e} \quad (25)$$

$$\approx \frac{a}{1+e}$$

a = new circular orbit. So:

$$\frac{4}{3} \frac{\langle \underline{S}_r \cdot \underline{S}_r \rangle}{a^2} e = \frac{6\pi MG}{ac^2(1-e)} \quad (25)$$